

EMERGENCY EVACUATION ILLUMINATION

1. INTRODUCTION: The purpose of this Aerospace Recommended Practice (ARP) is to provide criteria which will lead to standards of illumination for emergency evacuation in passenger or cargo transport aircraft such that the emergency illumination will facilitate egress under emergency conditions.

Consideration is given to existing requirements of the FAA and to the recommendations of aircraft operators and those involved in the manufacture or use of the Emergency Lighting System. Occupant safety is the primary objective, with appropriate provisions for crew system control taken into consideration. The criteria established herein are intended to produce an Emergency Lighting System which will comply with The Federal Regulations for Commercial Transport Airplanes. However, these recommendations are but one means of meeting the objective. Alternate practices may provide equivalent or superior results.

2. PURPOSE:

- 2.1 Definition: Emergency illumination is that illumination which is provided when normal illumination is unavailable. Emergency illumination should be adequate to permit aircraft occupants to locate, proceed to operate, and use cabin emergency exits, escape slides, life jackets, life rafts, slide rafts and special survival equipment.
- 2.2 Scope: This ARP provides criteria for design and location of power supplies, controls, light fixtures, and associated equipment which are used to provide emergency illumination in transport aircraft, designed to FAR 25 for operation under FAR 91 and FAR 121.

It is not the purpose of ARP to specify design methods to be followed in the accomplishment of the stated objectives.

3. DETAIL RECOMMENDATIONS:

- 3.1 General Provisions:

SAE Technical Board rules provide that: "All technical reports, including standards approved and practices recommended, are advisory only. Their use by anyone engaged in industry or trade or their use by governmental agencies is entirely voluntary. There is no agreement to adhere to any SAE standard or recommended practice, and no commitment to conform to or be guided by any technical report. In formulating and approving technical reports, the Board and its Committees will not investigate or consider patents which may apply to the subject matter. Prospective users of the report are responsible for protecting themselves against liability for infringement of patents."

- 3.1.1 System Description: An emergency lighting system(s) independent of the main lighting system, should be provided for (1) interior emergency lighting which includes illumination of the cabin, exit areas, floors of the passageways leading to each passenger emergency exit, (2) exit locating and marking signs, and (3) exterior emergency lighting.
- 3.1.2 Dark Adaptation: Emergency illumination should be designed to minimize glare or not affect dark adaptation in such a way as to compromise the ability to escape.
- 3.1.3 Obstruction of Light: Emergency light fixtures should be in locations which assure that devices such as open life raft compartment doors or emergency equipment doors do not obstruct the illumination of emergency equipment and escape routes.
- 3.1.4 Crash Protection: The emergency illumination system should be designed, installed as specified in 3.1.10 and located in a manner that will minimize damage to or loss of any portion of the emergency illumination as a result of a survivable emergency landing on land or water.
- 3.1.5 Fuselage Break-up: Break-up of the fuselage should not render inoperative more than 25% of the required electrically illuminated emergency lights in addition to those directly destroyed by the break. Exterior emergency light at each useable exit and each exit marking sign should remain operative exclusive of those that are directly damaged by the separation.

In the case of a combination passenger cargo aircraft, this standard need not be applied to the remaining passenger compartment providing the lighting for the total fuselage, if it were all passenger, meets the 25 percent criteria.

- 3.1.6 Operating Duration: Both interior and exterior emergency illumination should be maintained at or above minimum levels for no less than ten minutes at critical ambient conditions after an emergency landing.

Critical ambient conditions should be determined for each application and utilized for testing to satisfy this recommendation. If battery packs are used conditions to be considered are: cold soak at lowest ambient temperature anticipated, cabin heating for an appropriate period, passenger loading, taxi, and emergency evacuation during taxi and/or take-off run. If battery packs are used and installed outside the heated and/or pressurized compartment the cabin heating period may not be appropriate; the landing phase after a long cruise may be more critical.

- 3.1.7 Control and Indication: The emergency lighting system should provide for manual operation of the lights from the flight crew station and from a location in the passenger compartment that is within reach of a flight attendant seated in an assigned seat. In an emergency lighting system design that allows the activation of flight deck emergency lighting from the cabin, such facilities should be available only when normal aircraft power is off. Emergency lighting in the flight deck should be arranged so as to minimize glare and problems with dark adaptation by the flight crew.

- 3.1.7.1 The cockpit control device should have ON, OFF, and ARMED positions and have means to safeguard against inadvertent operation of the control device from the ARMED and from the ON positions. When either the cockpit control or the control at the cabin attendant's seat is ON the lights should remain lighted upon interruption of the airplane's normal electrical power source, except as noted in 3.1.5. When the cockpit control is in the armed position, the lights should illuminate upon interruption of the airplane's normal electrical power sources and remain illuminated except for an interruption caused by a transverse separation of the fuselage during a crash landing. Operation of the emergency lighting system should not be dependent upon normal aircraft power.
- 3.1.7.2 The cabin control device should have a means to safeguard against inadvertent operation and be capable of turning on the emergency light system, even with the cockpit control device in the OFF or ARMED position.
- 3.1.7.3 There should be a flight crew warning light which illuminates when normal aircraft power is on in the airplane and the emergency lighting system is not ARMED.
- 3.1.8 Other Use of Emergency Lighting System Components: Illuminated signs and other portions of the emergency lighting system may be used under normal conditions, provided that depletion of the emergency illumination power supply is not possible, beyond the requirement of Paragraph 3.1.6.
- 3.1.9 Independent Power Source: If rechargeable batteries are used as the energy supply for the emergency lighting system, they may be recharged from the airplane's main electric power system: provided, that, the charging circuit is designed to preclude inadvertent battery discharge into charging circuit faults. It is desirable that the charging system be capable of recharging within one hour to facilitate aircraft dispatch. However, achievement of this charging rate should not reduce the system integrity.
- 3.1.10 Inertia Forces: Components of the emergency lighting system, including batteries, wiring, relays, lamps, and switches should be capable of normal operation after having been subjected to the inertia forces acting separately relative to the surrounding structure: (a) upward 2.0 g; (b) forward 9.0 g; (c) sideward 1.5 g; (d) downward 4.5 g.
- Design consideration should be given to the fact that passenger survival may occur in crashes experiencing 20 g upward, forward and downward directions and 10 g sideways.
- 3.1.11 Functional Testing: As a design objective, the emergency lighting system and associated equipment should be designed and installed so that functional tests of the system can be readily performed.

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3.1.12 Maintainability: The emergency lighting system and equipment should be designed and qualified for ease of maintainability and highly reliable operation under the applicable aircraft environments.

## 3.2 Interior Illumination:

### 3.2.1 General Emergency Illumination:

3.2.1.1 Sufficient general emergency illumination should be provided in all compartments normally occupiable during taxi, takeoff and landing to permit occupants to don life jackets, to operate escape means, and to avoid obstacles while moving toward exits.

3.2.1.2 White flood light should be provided on passenger and cargo aircraft, as applicable:

- (a) 64 cm (25 inches) above floor at exit door in flight crew station of not less than 0.54 lux (0.05 footcandle).
- (b) Average illumination of not less than 0.54 lux (0.05 footcandle) along the center of the main passenger aisle(s) and cross aisle(s) between main aisles. This average should be determined from measurements made every 102 cm (40 inches) along the center of the main passenger aisle(s) at seat arm rest height. The illumination of each 102 cm (40 inch) interval should not be less than 0.11 lux (0.01 footcandle) at seat arm rest height measured parallel to the floor.
- (c) The floor of the passageway leading to each floor level passenger emergency exit, between the main aisles and the exit openings, should be provided with illumination that is not less than 0.22 lux (0.02 footcandle) measured along a line that is within 15 cm (six inches) of and parallel to the floor and centered on the passenger evacuation path.

3.2.1.3 For aircraft with passenger compartments on more than one deck level intended for occupancy during taxi, takeoff or landing, the recommendation of Sections 3.1 and 3.2 should apply to each deck level and interconnecting stairways.

3.2.1.4 (Reserved for lighting for dense smoke in cabin - subject to testing)

### 3.2.2 Exit Sign Illumination:

3.2.2.1 Location of each passenger exit should be indicated by a sign(s) visible to occupants approaching along the main passenger aisle(s).

- (a) Exit locator signs should be located above the aisle(s) near each exit and be internally electrically illuminated.
- (b) An exit marking sign should be located next to each exit and be internally electrically illuminated.

## 3.2.2.1 (Continued)

- (c) Exit signs should be located on each bulkhead or divider that prevents fore and aft vision along the passenger cabin to indicate emergency exits beyond and obscured by the bulkhead or divider.
  - (d) Stairway location should be indicated by an exit sign visible to passengers approaching the main aisle(s) and be internally electrically illuminated.
- 3.2.2.2 For signs specified in 3.2.2.1 (a), (b) and (c) and (d) the lighted background-to-letter contrast should be at least 10:1. This value is also applicable to symbols, arrows and instructional placards.
- 3.2.2.3 Lettering of the word "EXIT" on exit signs should be red and be a minimum of 381 mm (1-1/2 inches) high on illuminated white background. The letter height to stroke width ratio should not be more than 7:1 nor less than 6:1. The background should have an area of at least 135.4 square cm (21 square inches) excluding the letters, arrows and symbols.
- 3.2.2.4 The background luminance of lighted areas of exit signs per paragraph 3.2.2.1 (a) and (b) should be no less than 86 cd/m<sup>2</sup> (25 foot lamberts) and the ratio of maximum to minimum luminance should be no greater than 3:1.
- 3.2.2.5 The initial luminance of the exit locator signs per paragraph 3.2.2.1 (c) and (d) should be no less than 1.3 cd/m<sup>2</sup> (0.38 foot lambert). The colors may be reversed in case of a sign that is self-illuminated by other than electrical means.
- 3.2.2.6 For self-illuminated signs, the non-lighted contrast ratio under ambient light between the intelligence and background should be no less than 3.0.

3.2.3 Instructional Placards and Signs:

- 3.2.3.1 The location of the exit opening handle and instruction for opening the exit should be illuminated.
- 3.2.3.2 Life raft and escape assist means stowage locations should be marked by illuminated placards or signs, with instructions as to how to operate the compartment doors. Refer to ARP 577 for additional placard recommendations.
- (a) All letters should be no less than 7.9 mm (5/16 inch) high and the luminance of lighted areas, whether floodlighted by either emergency area lighting, or electrically illuminated or self-illuminated, should be no less than 0.5 cd/m<sup>2</sup> (0.15 footlambert).
  - (b) If deployment and erection of the assist means is fully automatic with opening of the associated exit, illumination of the stowage locations is not necessary.

3.2.3.3 On aircraft with an overhead escape hatch emergency lighting should be provided to illuminate the hatch outline, release handle, signs and placards per 3.2.3.2. The hatch outline should be measured at not less than four points around the outline. Obstructions, such as bunks or seat backs, along the escape route should be similarly illuminated.

### 3.3 Exterior Illumination:

3.3.1 General Emergency Illumination: Evacuation demonstrations have revealed that evacuees whose vision is adapted to the relatively high intensity of normal cabin lighting may not have adequate time to adapt to the dark conditions beyond the deboarding end of the escape assist means. Design consideration should be given to providing sufficient illumination of the area beyond the assist means to permit evacuees to expeditiously move away from the exit area while avoiding inhospitable terrain, aircraft components or other objects. Levels of illumination and areas of coverage should be determined for each application.

3.3.2 Inflatable Readiness Indicators: At all external exits which are equipped with escape assist means that incorporate readiness indicators, such indicators should be adequately illuminated by the exterior emergency lighting systems so as to be visible from within the cabin prior to the initiation of an evacuation.

### 3.3.3 Overwing Exits:

3.3.3.1 Provide a minimum of 0.32 lux (0.03 footcandle) of illumination (measured normal to the direction of the incident light) in a 0.2 square meter (two-square-foot) area where the egressing person will make his first step outside the overwing exit(s). This light source may be inside the airplane. The 0.2 square meter (two-square-foot) area may include any steps that are installed to decrease the height of the first step from the exit to the wing, and the lighted area on the wing surface shall be considered to begin at the inboard edge of the shadow of the step, if the lighting is installed above the door and inside the airplane.

3.3.3.2 Provide a minimum of 0.54 lux (0.05 footcandle) of illumination (measured normal to the direction of the incident light) for a minimum width of 107 cm (42 inches) for a Type A overwing emergency exit and of 61 cm (two feet) for all other overwing emergency exits along the 30 percent of the slip-resistant portion of the escape route that is farthest from the exit.

3.3.3.3 Provide a minimum of 0.32 lux (0.03 footcandle) of illumination (measured normal to the direction of the incident light) on the ground surface with the airplane in each of the attitudes corresponding to the normal ground attitude and the collapse of one or more legs of the landing gear where an evacuee using the established escape route would normally first contact with the ground. If the escape route is over the wing trailing edge flaps