

Gas Turbine Engine Performance Presentation
for Digital Computer Programs Using FORTRAN 77

RATIONALE

As ARP4191C contains information that was incorporated from AS681, the S15 committee decided to upgrade ARP4191C to AS4191. Therefore as AS4191 supersedes ARP4191C, it (ARP4191C) should be cancelled.

CANCELLATION NOTICE

This document has been declared "CANCELLED" as of May 2009 and has been superseded by AS4191. By this action, this document will remain listed in the Numerical Section of the Aerospace Standards Index noting that it is superseded by AS4191.

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1. SCOPE:

This SAE Aerospace Recommended Practice (ARP) provides a method for digital computer programs for gas turbine engine performance, steady-state or transient, performance to be written using the FORTRAN 77 Language.

When it is agreed between the program User and Supplier that a particular program shall be supplied in FORTRAN 77, it is recommended that this ARP be used in conjunction with AS681 for steady-state and transient programs.

This ARP also describes how to take advantage of the FORTRAN 77 CHARACTER storage to extend the information interface between the calling program and the engine subroutine.

The ARP has the same major section numbers as AS681 to facilitate its use with this document. The information given in each section of this ARP is additional to that given in AS681.

2. REFERENCES:

2.1 Applicable Documents:

The following publications form a part of this specification to the extent specified herein. The latest issue of SAE publications shall apply. The applicable issue of other publications shall be the issue in effect on the date of the purchase order. In the event of conflict between the text of this specification and references cited herein, the text of this specification takes precedence. Nothing in this specification, however, supersedes applicable laws and regulations unless a specific exemption has been obtained.

2.1.1 SAE Publications: Available from SAE, 400 Commonwealth Drive, Warrendale, PA 15096-0001.

AS681	Gas Turbine Engine Steady-State and Transient Performance Presentation for Digital Computer Programs
AS755	Aircraft Propulsion System Performance Station Designation and Nomenclature

3. GENERAL REQUIREMENTS:

3.1 Program Operation:

The program will have the capability of functioning both as a subroutine (herein called the engine subroutine) and as an independent program (consisting of a MAIN program containing a CALL statement to the engine subroutine).

4. PROGRAMMING PRACTICES:

4.1 Program Language:

'FORTRAN 77' when used in this document refers to ANSI X3.9 1978.

Fixed character data shall be held in the labeled CHARACTER COMMONS CHRIN and CHROUT.

The program may provide an extra text interface in the CHARACTER COMMON CHREXP which contains information that is not available from CHRIN or CHROUT. The storage defined in CHREXP is quite large and, therefore, use of CHREXP must be coordinated between user and supplier.

(See Section 6 for details of CHRIN, CHROUT and CHREXP).

5. PROGRAM CAPABILITIES:

5.1 Ram Pressure Recovery:

SERAM options 3, 5, and 6, which provide for a functional relationship between ram recovery and engine airflow and/or freestream Mach number, require a user supplied subroutine with the following call sequence:

```
CALL ERAMX (XM, WR*, SECALC, ERAM)
```

*W1R or W1AR as appropriate.

6. INPUT/OUTPUT:

6.1 Program Interface Definition:

- 6.1.1 The communication between the calling program and the engine subroutine will be handled completely by nine labeled COMMONS. These will be FIXIN, VARIN, EXPIN and CHRIN (character input) for input and FIXOUT, VAROUT, EXPOUT, CHROUT (character output), and CHREXP (extra character output) for output. The engine subroutine will never store computed values in FIXIN, VARIN, EXPIN, or CHRIN. Both INTEGER and REAL items in the common blocks have the length of a single precision full word. This restriction does not apply to character storage as it is independent of the length of a byte or word on any particular computer.

- 6.1.2 The fixed input parameters placed in labeled common FIXIN will be limited to the order and quantity defined in 6.2. The fixed output (FIXOUT) will be limited to the order and quantity of parameters defined in 6.3. If any parameter in the FIXIN or FIXOUT list is not used for a particular application, then that location in common will be unused so that the parameters which follow it may be found in the locations indicated in the tables. The remaining required input and output will be dependent on engine configuration and contained in labeled commons VARIN and VAROUT, respectively. Labeled commons EXPIN and EXPOUT will be reserved for additional parameters available through coordination between the program supplier and user.
- 6.1.3 The nomenclature utilized in the fixed input and output common blocks (FIXIN and FIXOUT) of 6.2 and 6.3 is for clarification only. Logically equivalent nomenclature consistent with AS755 is permitted, e.g., W1 may be substituted for W1A in FIXOUT when multiple streams have not been distinguished.
- 6.1.4 Some output parameters have corresponding inputs in FIXIN, where their names are prefixed by the letter Z. The output locations in FIXOUT should contain the values of each parameter as used in the computation, irrespective of whether they have been specified in the input or overridden by the program.

6.2 Input:

- 6.2.1 The Fixed Input Labeled Common Block FIXIN: The fixed sequence list of the parameters and the identity of these parameters (with typical nomenclature consistent with AS755) are as follows: (all parameters are REAL floating point numbers, except as indicated)

1. NIN	Input file number (INTEGER)
2. NOUT	Output file number (INTEGER)
3. IND	Engine program indicator (INTEGER)
4. thru 21.	Reserved for historical consistency
22. CASE	Numerical case identification
23. ZALT	Geopotential pressure altitude
24. ZDTAMB	Ambient temperature minus standard Atmospheric temperature
25. ZDT1A	Temperature to be added to T1A
26. ZERM1A	Ram pressure recovery at station 1A
27. ZPWXH	Customer high pressure rotor power extraction
28. ZPAMB	Ambient pressure
29. ZPC	Power code
30. ZPLA	Power lever angle
31. ZP1A	Engine inlet total pressure at station 1A
32. ZRC	Rating code
33. SERAM	Ram pressure recovery selection
	Average Options
	SERAM = 1, Selects specified ram pressure recovery
	SERAM = 2, Selects input value of ram pressure recovery
	SERAM = 3, Selects ram pressure recovery from user supplied subroutine (ERAMX)
	Differentiated Options

6.2.1 (Continued):

	SERAM = 4, Selects input values of primary and secondary ram pressure recovery
	SERAM = 5, Selects input value of primary stream ram pressure recovery and calls user supplied subroutine (ERAMX) for secondary stream ram pressure recovery
	SERAM = 6, Selects primary and secondary stream ram pressure recoveries from user supplied subroutine (ERAMX)
34. SIM	Inlet Mode Selection SIM = 1, Selects altitude and Mach number SIM = 2, Selects pressure and temperatures SIM = Other than 1 or 2 coordinated between user and supplier
35. ZTAMB	Ambient temperature
36. ZT1A	Engine inlet total temperature at station 1A
37. ZWB3	High pressure compressor discharge bleed flow rate
38. ZWB3Q	High pressure compressor bleed flow ratio (discharge over component inlet)
39. ZXM	Free stream Mach number
40. ZERAM1	Primary stream ram pressure recovery
41. ZERM11	Secondary stream ram pressure recovery
42. --	Reserved for historical consistency
43. --	Reserved for historical consistency
44. --	Reserved for historical consistency
45. --	Reserved for historical consistency
46. SDIST	Inlet pressure and temperature distortion selection
47. FYPH	Primary maximum response frequency
48. FYSH	Secondary maximum response frequency
49. ZPWSD	Specified shaft power
50. ZTIME	Time from start of transient case
51. TIMEF1	Time at which frequency is changed to FYSH
52. TIMEF2	Time at which frequency reverts to FYPH
53. TIMEO	Output time interval
54. ZTIMET	Termination time of transient case
55. ZXJPTL	Polar moment of inertia of power turbine load
56. ZXNSD	Specified shaft rotational speed
57. ZTRQSD	Specified shaft torque
58. SWIND	Windmilling selection

6.2.2 The Variable Input Labeled Common Block VARIN: Will contain input dependent on engine configuration.

6.2.3 The Expanded Input Labeled Common Block EXPIN: Will contain input available through coordination between program supplier and user.

6.2.4 The Character Fixed Input Labeled Common Block CHRIN: Will contain

1. CTITLE User Title - CHARACTER * 72

6.3 Output:

6.3.1 The Fixed Output Labeled Common Block FIXOUT: The fixed sequence list of the parameters and the identity of these parameters (with typical nomenclature consistent with AS755) are as follows: (all parameters are REAL floating point numbers except as indicated).

- | | |
|--------------|---|
| 1. thru 42. | Reserved for historical consistency |
| 43. thru 52 | |
| NSI (1) thru | |
| NSI (10) | Numerical Status Indicator (DIMENSION 10 INTEGER) |
| 53. AE8 | Primary exhaust nozzle throat effective area |
| 54. AE18 | Bypass exhaust nozzle throat effective area |
| 55. ANGBT | Boat-tail angle |
| 56. FRAM | Ram drag |
| 57. FG | Gross thrust |
| 58. FGI | Ideal gross thrust |
| 59. FG19 | Bypass stream gross thrust |
| 60. FGI19 | Bypass stream ideal gross thrust |
| 61. FHV | Fuel lower heating value |
| 62. FN | Net thrust |
| 63. PB3 | High pressure compressor discharge bleed flow total pressure |
| 64. P7 | Primary exhaust flow total pressure |
| 65. P17 | Bypass exhaust flow total pressure |
| 66. SFC | Specific fuel consumption |
| 67. -- | Reserved for historical consistency |
| 68. TB3 | High pressure compressor discharge bleed flow total temperature |
| 69. TC | Control temperature (cockpit display) |
| 70. T7 | Primary exhaust flow total temperature |
| 71. T17 | Bypass exhaust flow total temperature |
| 72. WFE | Engine fuel flow rate |
| 73. WFT | Total fuel flow rate |
| 74. W1A | Engine inlet flow rate at station 1A |
| 75. W7 | Primary exhaust flow rate |
| 76. W17 | Bypass exhaust flow rate |
| 77. W2 | High pressure compressor inlet flow rate
(The full number representing the relevant station designation, e.g., W21, W215, W2A, will be defined by the program supplier.) |
| 78. XNH | High pressure rotor rotational speed |
| 79. XNI | Intermediate pressure rotor rotational speed |
| 80. XNL | Low pressure rotor rotational speed |
| 81. XNSD | Delivered shaft rotational speed |
| 82. ALT | Geopotential pressure altitude |