



# AEROSPACE MATERIAL SPECIFICATION

AMS6362

REV. G

Issued 1942-06  
Revised 2014-06

Superseding AMS6362F

Steel Tubing, Seamless  
0.95Cr - 0.20Mo (0.28 - 0.33C) (SAE 4130)  
150 ksi (1034 MPa) Tensile Strength  
(Composition similar to UNS G41300)

## RATIONALE

AMS6362G results from a Five Year Review and update of this specification.

### 1. SCOPE

#### 1.1 Form

This specification covers an aircraft-quality, low-alloy steel in the form of round, seamless tubing having a wall thickness not greater than 0.188 inch (4.78 mm).

#### 1.2 Application

This tubing has been used typically for general use where a minimum tensile strength of 150 ksi (1034 MPa) is required, but usage is not limited to such applications.

### 2. APPLICABLE DOCUMENTS

The issue of the following documents in effect on the date of the purchase order forms a part of this specification to the extent specified herein. The supplier may work to a subsequent revision of a document unless a specific document issue is specified. When the referenced document has been cancelled and no superseding document has been specified, the last published issue of that document shall apply.

#### 2.1 SAE Publications

Available from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001, Tel: 877-606-7323 (inside USA and Canada) or 724-776-4970 (outside USA), [www.sae.org](http://www.sae.org).

AMS2253 Tolerances, Carbon and Alloy Steel Tubing

AMS2259 Chemical Check Analysis Limits, Wrought Low-Alloy and Carbon Steels

AMS2301 Steel Cleanliness, Aircraft Quality, Magnetic Particle Inspection Procedure

AMS2370 Quality Assurance Sampling and Testing, Carbon and Low-Alloy Steel Wrought Products and Forging Stock

SAE Technical Standards Board Rules provide that: "This report is published by SAE to advance the state of technical and engineering sciences. The use of this report is entirely voluntary, and its applicability and suitability for any particular use, including any patent infringement arising therefrom, is the sole responsibility of the user."

SAE reviews each technical report at least every five years at which time it may be revised, reaffirmed, stabilized, or cancelled. SAE invites your written comments and suggestions.

Copyright © 2014 SAE International

All rights reserved. No part of this publication may be reproduced, stored in a retrieval system or transmitted, in any form or by any means, electronic, mechanical, photocopying, recording, or otherwise, without the prior written permission of SAE.

TO PLACE A DOCUMENT ORDER: Tel: 877-606-7323 (inside USA and Canada)  
Tel: +1 724-776-4970 (outside USA)  
Fax: 724-776-0790  
Email: [CustomerService@sae.org](mailto:CustomerService@sae.org)  
SAE WEB ADDRESS: <http://www.sae.org>

SAE values your input. To provide feedback on this Technical Report, please visit <http://www.sae.org/technical/standards/AMS6362G>

AMS2807 Identification, Carbon and Low-Alloy Steels, Corrosion and Heat-Resistant Steels and Alloys, Sheet, Strip, Plate and Aircraft Tubing

## 2.2 ASTM Publications

Available from ASTM International, 100 Barr Harbor Drive, P.O. Box C700, West Conshohocken, PA 19428-2959, Tel: 610-832-9585, [www.astm.org](http://www.astm.org).

ASTM A 370 Mechanical Testing of Steel Products

ASTM E 112 Determining Average Grain Size

ASTM E 350 Chemical Analysis of Carbon Steel, Low-Alloy Steel, Silicon Electrical Steel, Ingot Iron, and Wrought Iron

ASTM E 384 Knoop and Vickers Hardness of Materials

ASTM E 1444 Magnetic Particle Testing

## 3. TECHNICAL REQUIREMENTS

### 3.1 Composition

Shall conform to the percentages by weight shown in Table 1, determined by wet chemical methods in accordance with ASTM E 350, by spectrochemical methods, or by other analytical methods acceptable to purchaser:

TABLE 1 - COMPOSITION

Element	min	max
Carbon	0.28	0.33
Manganese	0.40	0.60
Silicon	0.15	0.35
Phosphorus	--	0.025
Sulfur	--	0.025
Chromium	0.80	1.10
Molybdenum	0.15	0.25
Nickel	--	0.25
Copper	--	0.35

#### 3.1.1 Check Analysis

Composition variations shall meet the requirements of AMS2259.

### 3.2 Condition

Hardened by quenching from a temperature within the range 1500 to 1600 °F (816 to 871 °C) and tempered to meet the requirements of 3.4.1.

### 3.3 Fabrication

Tubing shall be produced by a seamless process. Any surface finishing operation applied to remove objectionable pits and surface blemishes shall be performed prior to final heat treatment. A light polish to improve surface appearance may be employed after final heat treatment.

### 3.4 Properties

Tubing shall conform to the following requirements; tensile testing shall be performed in accordance with ASTM A 370:

## 3.4.1 Tensile Properties

Shall be as shown in Table 2.

TABLE 2 - MINIMUM TENSILE PROPERTIES

Tensile Strength	150 ksi (1034 MPa)
Yield Strength at 0.2% Offset	135 ksi ( 931 MPa)
Elongation in 2 Inches (50.8 mm)	
Full Tube	10%
Strip	6%

3.4.1.1 Mechanical property requirements for product outside the range covered by Table 2 (See 1.1) shall be agreed upon between purchaser and producer.

## 3.4.2 Average Grain Size

Shall be ASTM No. 5 or finer, determined in accordance with ASTM E 112.

## 3.4.3 Decarburization

3.4.3.1 Tubing ordered ground, turned, or polished shall be free from decarburization on the ground, turned, or polished surfaces. Decarburization on tubing ID shall not exceed the maximum depth specified in Table 3.

3.4.3.2 Allowable decarburization of pierced billets, of tubing for redrawing, or of tubing ordered to specified microstructural requirements shall be as agreed upon by purchaser and vendor.

3.4.3.3 Tubing to which 3.4.3.1 or 3.4.3.2 is not applicable shall be free from complete (ferritic) decarburization. Partial decarburization shall not exceed the limits specified in Table 3.

TABLE 3A – MAXIMUM DEPTH OF DECARBURIZATION LIMITS, INCH/POUND UNITS

Nominal Wall Thickness (T) Inch	Depth of Partial Decarburization	Depth of Partial Decarburization	Depth of Partial Decarburization
	Inch ID	Inch OD	Inch ID+OD
Up to 0.040, incl	0.18T	0.18T	0.25T
Over 0.040 to 0.050, incl	0.009	0.009	0.012
Over 0.050 to 0.070, incl	0.010	0.010	0.014
Over 0.070 to 0.080, incl	0.012	0.012	0.016
Over 0.080 to 0.090, incl	0.014	0.014	0.018
Over 0.090 to 0.100, incl	0.015	0.015	0.020
Over 0.100 to 0.150, incl	0.017	0.017	0.022
Over 0.150 to 0.188, incl	0.020	0.020	0.026

TABLE 3B – MAXIMUM DEPTH OF DECARBURIZATION LIMITS, SI UNITS

Nominal Wall Thickness (T) Millimeters	Depth of Partial Decarburization	Depth of Partial Decarburization	Depth of Partial Decarburization
	Millimeters ID	Millimeters OD	Millimeters ID+OD
Up to 1.02, incl	0.18T	0.18T	0.25T
Over 1.02 to 1.27, incl	0.23	0.23	0.30
Over 1.27 to 1.78, incl	0.25	0.25	0.36
Over 1.78 to 2.03, incl	0.30	0.30	0.41
Over 2.03 to 2.29, incl	0.36	0.36	0.46
Over 2.29 to 2.54, incl	0.38	0.38	0.51
Over 2.54 to 3.81, incl	0.43	0.43	0.56
Over 3.81 to 4.78, incl	0.51	0.51	0.66

3.4.3.4 Decarburization shall be measured by the metallographic method or by microhardness traverse method in accordance with ASTM E 384. The microhardness method shall be conducted on a hardened, but untempered, specimen protected during heat treatment to prevent changes in surface carbon content. Depth of decarburization, when measured by the microhardness method, is defined as the perpendicular distance from the surface to the depth under that surface where there is no further increase in hardness. Such measurements shall be far enough away from any adjacent surface to be uninfluenced by decarburization thereon. In case of dispute, the depth of decarburization determined via the microhardness traverse method shall govern.

3.4.3.4.1 When determining the depth of decarburization, it is permissible to disregard local areas provided the decarburization of such areas does not exceed the above limits by more than 0.005 inch (0.13 mm) and the width is 0.065 inch (1.65 mm) or less.

### 3.5 Quality

Tubing, as received by purchaser, shall be uniform in quality and condition and shall have a finish conforming to the best practice for high quality aircraft tubing. It shall be smooth and free from heavy scale or oxide, burrs, seams, tears, grooves, laminations, slivers, pits, and other imperfections detrimental to usage of the tubing. Surface imperfections such as handling marks, straightening marks, light mandrel and die marks, shallow pits, and scale pattern will not be considered injurious if the imperfections are removable within the tolerances specified for wall thickness but removal of such imperfections is not required.

3.5.1 Steel shall be aircraft quality conforming to AMS2301.

3.5.2 When specified, the tubing, either with or without machining of the surface, shall be subjected to magnetic particle inspection in accordance with ASTM E 1444. Standards for acceptance shall be as agreed upon by purchaser and vendor.

### 3.6 Tolerances

Shall be in accordance with AMS2253.

## 4. QUALITY ASSURANCE PROVISIONS

### 4.1 Responsibility for Inspection

The vendor of tubing shall supply all samples for vendor's tests and shall be responsible for the performance of all required tests. Purchaser reserves the right to sample and to perform any confirmatory testing deemed necessary to ensure that the tubing conforms to specified requirements.

### 4.2 Classification of Tests

All technical requirements are acceptance tests and shall be performed on each heat or lot as applicable.

### 4.3 Sampling and Testing

Shall be in accordance with AMS2370.