

AERONAUTICAL MATERIAL SPECIFICATIONS

AMS 5570G

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STEEL TUBING, SEAMLESS, CORROSION AND HEAT RESISTANT
18Cr - 11Ni - Ti (SAE 30321)

1. ACKNOWLEDGMENT: A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
2. APPLICATION: Parts and assemblies requiring both corrosion and heat resistance, especially when such parts and assemblies are welded during fabrication. Parts and assemblies requiring oxidation resistance up to approximately 1500 F, but useful at that temperature only when stresses are low.

3. COMPOSITION:

		Check Analysis	
		Under Min	or Over Max
Carbon	0.08 max	--	0.01
Manganese	2.00 max	--	0.04
Silicon	0.40 - 1.00	0.05	0.05
Phosphorus	0.040 max	--	0.005
Sulfur	0.030 max	--	0.005
Chromium	17.00 - 20.00	0.20	0.20
Nickel	9.00 - 13.00	0.15	0.15
Titanium	6xC - 0.70	0.05	0.05
Molybdenum	0.50 max	--	0.03
Copper	0.50 max	--	0.03

4. CONDITION: Solution heat treated and descaled.

- 4.1 Fabrication: Any surface finishing operation applied to remove objectionable pits and surface blemishes shall be performed prior to the last solution heat treatment. A light polish to improve surface appearance may be employed after solution heat treatment. Passivation treatment shall follow any polishing treatment.

5. TECHNICAL REQUIREMENTS:

5.1 Tensile Properties:

Nominal OD Inches	Wall Thickness Inch	Tensile Strength psi, max	Elongation % in 2 in., min	
			Strip	Full Tube
0.188 and under	0.016 and under	120,000	--	33
	Over 0.016	105,000	--	35
Over 0.188 to 0.500, incl	0.010 and under	115,000	30	35
	Over 0.010	105,000	30	35
Over 0.500	0.010 and under	120,000	25	30
	Over 0.010	105,000	30	35

Section 7C of the SAE Technical Board rules provides that: "All technical reports, including standards approved and practices recommended, are advisory only. Their use by anyone engaged in industry or trade is entirely voluntary. There is no requirement to adhere to any SAE standard or recommended practice, and no commitment to conform to or be guided by any technical report. In formulating and approving technical reports, the Board and its Committees will not investigate or consider patents which may apply to the subject matter. Prospective users of the reports are responsible for protecting themselves against infringement of patents."

- 5.2 Flarability: Tubing shall be capable of being flared without formation of cracks or other visible defects. Specimens for flaring may be cut from any portion of the tube, or an entire tube may be used as a specimen. The end of the specimen to be flared shall be cut square, with the cut end smooth and free from burrs, but not rounded. The specimen shall, at room temperature, be forced axially with steady pressure over a hardened and polished tapered steel pin having a 74 deg included angle, to produce a flare having the permanent expanded OD specified in the following table.

Nominal OD Inches	Expanded OD Inches, min	Nominal OD Inches	Expanded OD Inches, min
0.125	0.200	0.750	0.937
0.188	0.290	1.000	1.187
0.250	0.359	1.250	1.500
0.312	0.421	1.500	1.721
0.375	0.484	1.750	2.106
0.500	0.656	2.000	2.356
0.625	0.781		

- 5.2.1 Tubing with intermediate nominal OD shall take the same percentage flare as that for the next larger OD.

- 5.2.2 Tubing with nominal OD greater than 2.00 in. or less than 0.125 in. shall have flarability as agreed upon by purchaser and vendor.

- 5.3 Embrittlement: Tubing shall be capable of meeting the following test:

- 5.3.1 Test specimens, after being heated at 1200 F \pm 10 for 2 hr and air cooled, shall withstand immersion for 48 hr in a boiling aqueous solution containing 100 g of $\text{CuSO}_4 \cdot 5\text{H}_2\text{O}$ and 100 ml of H_2SO_4 (sp gr 1.84) per liter of solution under a reflux condenser, without evidence of intercrystalline surface attack. After such immersion, full cross-sectional specimens of tubing 0.625 in. or less in diameter shall be flattened to a total thickness under load of 3 times the wall thickness of the tubing, and 1 in. long specimens of tubing over 0.625 in. in diameter shall be split and bent 180 deg with outside surface of tube on inside of bend, around a diameter equal to the wall thickness, without showing evidence of cracks or defects. In either flattening or bending, the fold shall be made parallel to the axis of the tube.

6. QUALITY: Tubing shall have a good workmanlike finish conforming to the best practice for high quality aircraft material. Tubing shall be uniform in quality and condition, clean, sound, and free from grease or other foreign matter, and from internal and external imperfections detrimental to fabrication or to performance of parts.

7. TOLERANCES: Unless otherwise specified, tolerances shall conform to the latest issue of AMS 2243 as applicable. Diameter tolerances shall conform to Table I, columns headed "Annealed or Solution Heat Treated".

8. REPORTS:

- 8.1 Unless otherwise specified, the vendor of the product shall furnish with each shipment three copies of a report of the results of tests for chemical composition of each heat in the shipment and for tensile properties of each size from each heat. This report shall include the purchase order number, heat number, material specification number, size, and quantity from each heat.