

AERONAUTICAL MATERIAL SPECIFICATION

Society of Automotive Engineers, Inc.
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STEEL TUBING, WELDED
Low Carbon
Annealed

1. ACKNOWLEDGMENT: A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
2. TYPE: Electric-resistance-welded, shielded-arc-welded, or gas-welded.
3. APPLICATION: Primarily for parts requiring superior quality welded tubing suitable for severe forming and for welding and brazing.
4. COMPOSITION:

Ø	Check Analysis		
	Under Min or Over Max		
Carbon	0.13 max	--	0.03
Manganese	0.60 max	--	0.03
Phosphorus	0.01 max	--	0.01
Sulfur	0.05 max	--	0.01

5. CONDITION: Normalized or annealed.
 - 5.1 Fabrication: Any surface finishing operation applied to remove objectionable pits and surface blemishes shall be performed prior to the last annealing. A light polish to improve surface appearance may be employed after annealing.

6. TECHNICAL REQUIREMENTS:

6.1 Tensile Properties:

Nominal OD Inches	Elongation	
	% in 2 in., min	
	Full Tube	Strip
0.50 and under	32	--
Over 0.50 to 5.50, incl	35	25

- 6.2 Crushing Test: One sample or more shall be selected from each 1000 ft or less from each lot of tubing in the shipment. Test specimens shall have a length equal to 1-1/2 times the nominal outside diameter and shall withstand crushing lengthwise under a gradually applied load until the cross sectional dimension is increased in one zone by 20%, or until one complete fold is formed, or until the specimen is reduced in length to 2/3 of the original length, without failure of the weld occurring.

Section 7C of the SAE Technical Board rules provides that: "All technical reports, including standards approved and practices recommended, are advisory only. Their use by anyone engaged in industry or trade is entirely voluntary. There is no agreement to adhere to any SAE standard or recommended practice, and no commitment to conform to or be guided by any technical report. In formulating and approving technical reports, the Board and its Committees will not investigate or consider patents which may apply to the subject matter. Prospective users of the report are responsible for protecting themselves against liability for infringement of patents."

6.3 Flarability: Tubing shall be capable of being flared without formation of cracks or other visible defects. Specimens for flaring may be cut from any portion of the tube, or an entire tube may be used as a specimen. The end of the specimen to be flared shall be cut square, with the cut end smooth and free from burrs, but not rounded. The specimen shall, at room temperature, be forced axially with steady pressure over a hardened and polished tapered steel pin having a 74 degree included angle, to produce a flare having the permanent percentage OD increase specified in the following table:

Nominal Wall Thickness	% OD Increase
7% of OD and under	35 min
Over 7% of OD	45 min

6.4 Non-destructive Test: Each length of tubing shall be subjected to a non-destructive test by the tube manufacturer for the detection of injurious imperfections. The method of testing shall be capable of detecting all imperfections, interior and exterior, with a length greater than 1/16 in and a total depth equivalent to half the nominal wall thickness of the tube.

7. QUALITY:

7.1 Tubing shall be suitable for use in aircraft, shall be uniform in condition, and shall not reveal defects during fabrication processes.

7.2 Tubing shall have a good workmanlike finish conforming to the best practice for high quality aircraft material. It shall be smooth, clean, and free from heavy scale or oxide, burrs, seams, tears, grooves, laminations, slivers, pits, and other injurious defects. Surface imperfections such as handling marks, straightening marks, light mandrel and die marks, shallow pits, and scale pattern will not be considered as injurious defects, provided the imperfections are removable within the tolerances specified for diameter and wall thickness. The removal of surface imperfections is not required.

7.3 Steel used for manufacture of tubing shall be of a quality satisfactory for fabrication of parts which may be subjected to a method of inspection which will disclose injurious tubing defects as defined in 7.2.

8. TOLERANCES: Unless otherwise specified, tolerances shall conform to the latest issue of AMS 2253 as applicable to Aircraft Type - Welded. Straightness tolerance shall conform to Table VIII and flash tolerances shall conform to the following:

8.1 The outside surface shall be free from welding flash and the maximum height of the inside flash shall not exceed 0.007 inch.

9. REPORTS:

9.1 Unless otherwise specified, the vendor of tubing shall furnish with each shipment three copies of a report of the results of tests for chemical composition of each heat in the shipment and a statement that the tubing was tested and accepted under the non-destructive test. This report shall include the purchase order number, heat number, material specification number, size, and quantity from each heat.