

AEROSPACE

MATERIAL SPECIFICATIONS

AMS 4119B

SOCIETY OF AUTOMOTIVE ENGINEERS, Inc.

485 Lexington Ave., New York, N.Y. 10017

Issued 6-30-60

Revised 2-15-65

ALUMINUM ALLOY BARS, ROLLED OR COLD FINISHED

4.5Cu - 1.5Mg - 0.6Mn (2024-T351)

Stress-Relief Stretched

1. **ACKNOWLEDGMENT:** A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.
2. **FORM:** Bars and rods.
3. **APPLICATION:** Primarily for machined parts subject to excessive warpage during machining due to residual stresses, and for parts requiring good strength and whose fabrication does not involve welding. Certain design and processing procedures may cause this material to be susceptible to stress corrosion cracking; ARP 823 recommends practices to minimize such conditions.

4. **COMPOSITION:**

	min	max
Copper	3.8	4.9
Magnesium	1.2	1.8
Manganese	0.30	0.9
Iron	--	0.50
Silicon	--	0.50
Zinc	--	0.25
Chromium	--	0.10
Other Impurities, each	--	0.05
Other Impurities, total	--	0.15
Aluminum		remainder

5. **CONDITION:** Rolled or cold finished, solution heat treated, and stress-relieved by stretching.

- 5.1 Material shall be stretched in the solution heat treated condition to produce a nominal permanent set of 1-1/2%, but not less than 1% nor more than 3%.
- 5.2 Material shall receive no further straightening operations after stretching, unless specifically authorized.

6. **TECHNICAL REQUIREMENTS:**

- 6.1 **Tensile Properties:** Except as specified in 6.1.2, the following requirements apply to bars 0.500 in. \emptyset and over in least distance between parallel sides and to rods 0.500 in. and over in diameter:

Tensile Strength, psi	62,000 min
Yield Strength at 0.2% Offset or at 0.0116 in. in 2 in. Extension Under Load (E = 10, 500, 000), psi	40,000 min
Elongation, % in 2 in. or 4D	10 min

- 6.1.1 When a dispute occurs between purchaser and vendor over the yield strength value, yield strength determined by the offset method shall apply.