
**Aircraft ground support
equipment — Tow bars**

Équipement au sol pour aéronefs — Barres de tractage

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Contents

	Page
Foreword	iv
1 Scope	1
2 Normative references	1
3 Aircraft mass categories	1
4 Tow bar general requirements	2
5 Aircraft connection	2
5.1 Dimension of tow bar connection to aircraft	2
5.2 Aircraft interface requirements	2
5.3 Aircraft protective provisions	2
5.3.1 General requirements	2
5.3.2 Shear pins	3
5.3.3 Marking	4
6 Tractor connection	4
7 Tow bar undercarriage	5

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Foreword

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The procedures used to develop this document and those intended for its further maintenance are described in the ISO/IEC Directives, Part 1. In particular the different approval criteria needed for the different types of ISO documents should be noted. This document was drafted in accordance with the editorial rules of the ISO/IEC Directives, Part 2 (see www.iso.org/directives).

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For an explanation on the meaning of ISO specific terms and expressions related to conformity assessment, as well as information about ISO's adherence to the WTO principles in the Technical Barriers to Trade (TBT) see the following URL: [Foreword - Supplementary information](#)

The committee responsible for this document is ISO/TC 20, *Aircraft and space vehicles*, Subcommittee SC 9, *Air cargo and ground equipment*.

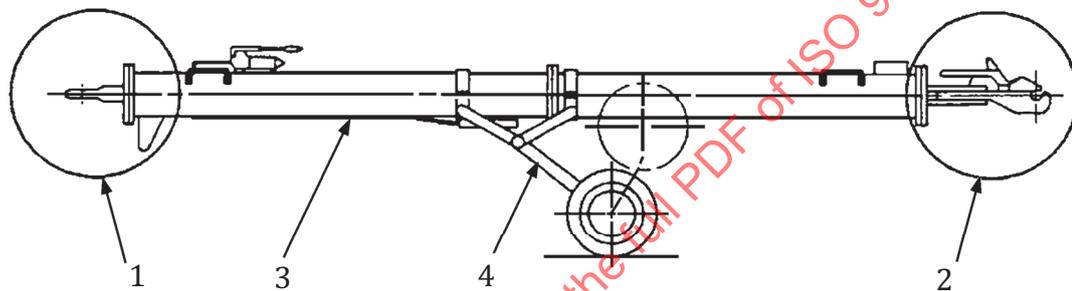
This second edition cancels and replaces the first edition (ISO 9667:1998), which has been technically revised.

Aircraft ground support equipment — Tow bars

1 Scope

This International Standard specifies dimensional and physical requirements of tow bar connections to tractor and aircraft (see [Figure 1](#)). It is applicable to any new type of commercial-transport-category aircraft tow bar designed or built after publication of this International Standard.

The purpose of this International Standard is to standardize tow bar attachments to airplane and tractor according to the mass category of the towed aircraft, so that one tow bar head with different shear levels can be used for all aircraft that are within the same mass category and are manufactured in compliance with ISO 8267 (all parts).



Key

- 1 tractor connection
- 2 aircraft connection
- 3 tow bar body
- 4 tow bar undercarriage

Figure 1 — Tow bar

2 Normative references

The following documents, in whole or in part, are normatively referenced in this document and are indispensable for its application. For dated references, only the edition cited applies. For undated references, the latest edition of the referenced document (including any amendments) applies.

ISO 8267-1, *Aircraft — Tow bar attachment fittings interface requirements — Part 1: Main line aircraft*

ISO 8267-2, *Aircraft — Tow bar attachment fittings interface requirements — Part 2: Regional aircraft*

EN 12312-7:2005+A1:2009, *Aircraft ground support equipment — Specific requirements — Part 7: Aircraft movement equipment*

3 Aircraft mass categories

The aircraft mass categories shall be in accordance with ISO 8267 (all parts).

A given tow bar type may be used on several aircraft types within the same ISO 8267-1 or ISO 8267-2, as appropriate, weight category, provided

- their nose landing gear towing interface meets the ISO 8267-1 or ISO 8267-2 requirements for that category, and
- the protective provisions are chosen to meet or exceed the minimum requirement for each aircraft.

4 Tow bar general requirements

As a safety feature the tow bar (body or head holding section or retaining feature) shall be designed to sustain at least 150 % of the protective device(s) functioning point or as otherwise specified by the customer.

The tow bar, when unladen, shall be both towable and stable at speeds up to 25 km/h (15 mph). The maximum towing speed shall be suitably placarded.

For use in the EU and EFTA, the tow bar shall, in addition, meet the Health and Safety requirements contained in EN 12312-7:2005, 5.9.

5 Aircraft connection

5.1 Dimension of tow bar connection to aircraft

The standard configuration of the tow bar connection to the aircraft shall be compatible with the horizontal pin of the aircraft tow bar attachment fitting as specified in ISO 8267 (all parts).

NOTE Several common aircraft types designed prior to publication of ISO 8267 (all parts) still use non-standard nose gear tow bar attachment fittings. Refer to the aircraft type manufacturer's documentation for interface requirements.

5.2 Aircraft interface requirements

The design of the tow bar aircraft connection device that clamps to the horizontal cylindrical pin of the aircraft

- shall grip the pin uniformly over 96 % to 98 % of its length,
- shall be designed to eliminate inadvertent disengagement of the tow bar during towing or pushing operations,
- shall be designed to provide adequate clearance during engagement and disengagement of the tow bar from the aircraft connection [allowable space envelope for clearance is specified in ISO 8267 (all parts)], and
- should be adjustable to provide pressure on the pin when locked.

NOTE Several common aircraft types designed prior to publication of ISO 8267 (all parts) still use non-standard nose gear tow bar attachment fittings. Refer to the aircraft type manufacturer's Aircraft Characteristics for Airport Planning Manual for interface requirements.

5.3 Aircraft protective provisions

5.3.1 General requirements

The tow bar should be so constructed as to prevent any item from becoming a Foreign Object Damage (FOD) hazard, e.g. broken parts of the shear pin and any bushes always remain captive to prevent aircraft engine ingress.

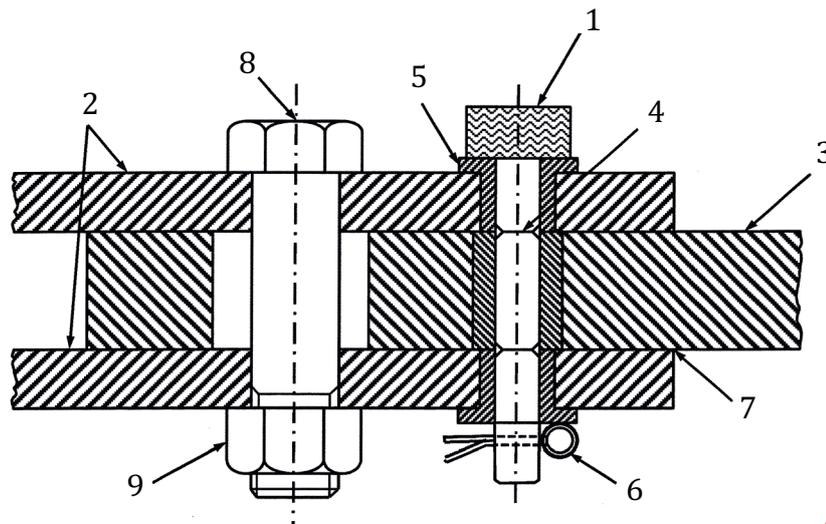
The tow bar shall incorporate a device or devices, such as for example shear pins (typically two, respectively for push/pull and torsion protection), to protect the aircraft from damage due to towing which shall

- relieve fore/aft and torsional towing forces applied to the aircraft nose gear through the tow bar which exceed the maximum force recommendations of the aircraft manufacturer by means of a shock absorbing device that is designed to function at values below tow bar design values,
- have an energy-absorbing device in order to limit loads induced in the landing gear that shall be designed to function at values beyond tow bar design values,
- simultaneously alert the tow tractor operator, audibly, visually or both, that a device has been actuated,
- be replaceable or repairable with a minimum of effort, and
- transfer overload to a retaining feature preventing separation and therefore loss of control of the aircraft from the towing vehicle. The retaining feature shall allow free rotation on $\pm 90^\circ$ about aircraft centreline at the retaining fastener. The retaining feature should be designed so that the towing forces do not exceed the aircraft type manufacturer's maximum force recommendations. See [Figure 2](#) example.

5.3.2 Shear pins

If shear pins are used as a protective device, they shall

- discourage hazardous replacement and encourage the exclusive use of manufacturer's original parts and be clearly identified for their specific usage,
- shear at or below the nominal force specified by the aircraft manufacturer(s),
- have a functioning point within a relative tolerance of (+0/-10 %), and
- have a specific predetermined shear plane, whether provided with or without hardened bushings (see [Figure 2](#)).



Key

- 1 shear pin
- 2 head retention plates
- 3 head
- 4 preset shear plane
- 5 hardened steel bushings
- 6 retaining clip
- 7 sliding fit
- 8 head retention bolt
- 9 self-locking nut

NOTE Do not clamp up assembly.

Figure 2 — Typical shear pin/bushing configuration

5.3.3 Marking

The tow bar shall be clearly and indelibly marked to identify the allowable type(s) of aircraft and maximum towing force.

6 Tractor connection

The attachment on the tow bar for connection to the tractor

- shall be as shown in [Figure 3](#) for all mass categories as specified in [Clause 3](#), and
- shall contain a rotating tow bar eye.